

Finding Drag Coefficient Using Solidworks Flow Simulation

An Analysis of Drag Coefficient at Hurricane Windspeeds from a Numerical Simulation of Dynamical Water Level Changes in Lake Okeechobee, FloridaA Free-flight Investigation of the Drag Coefficients of Two Single-engine Supersonic Interceptor Configurations from Mach Number 0.8 to 1.90 to Determine the Effect of Inlet and Engine LocationsVariation of the Drag Coefficient with Wind and Wave StateMultiphase Flows with Droplets and ParticlesDrag Coefficients for Spheres in Free Molecular Flow in O at Satellite VelocitiesThe Variation of the Drag Coefficient in the Marine Surface Layer Due to Temporal and Spatial Variations of the Surface Wind and Sea StateSummary of Drag Coefficients of Various Shaped CylindersDetermination of the Hypersonic-continuum/rarefield-flow Drag Coefficient of the Viking Lander Capsule 1 Aeroshell from Flight DataShock Tube Determination of the Drag Coefficient of Small Spherical ParticlesDrag coefficients of inert, burning, or evaporating particles accelerating in gas streamsDetermination of the Hypersonic-continuum/rarefied-flow Drag Coefficient of the Viking Lander Capsule 1 Aeroshell from Flight DataFree Wake Analysis of Hover Performance Using a New Influence Coefficient MethodA Two-dimensional Study of the Influence on Target Loading of Numerical Wave Reflections from Transmissive Computational BoundariesUniversity of Iowa Studies in EngineeringDrag Associated with Separated Flow Over Two-dimensional V-shaped Notches Under Transonic and Supersonic ConditionsThe AeroplaneAero DigestExperiments on Drag of Revolving Disks, Cylinders and Streamline Rods at High SpeedsFree-flight Drag Measurements of Rocket- Boosted Models of Two Reentry Body Configurations at Mach Numbers from 4.3 to 0.6ESDU Engineering Sciences Data: Drag of excrescences, undercarriage, canopy Robert E. Whitaker Joseph H. Judd Beverly J. Byars Clayton T. Crowe John W. Boring H. Michael Byrne C. F. Heddleson Robert C. Blanchard Bruce P. Selberg Clayton T. Crowe Todd R. Quackenbush Richard E. Lottero Ronald H. Howell Theodore Theodorsen An Analysis of Drag Coefficient at Hurricane Windspeeds from a Numerical Simulation of Dynamical Water Level Changes in Lake Okeechobee, Florida A Free-flight Investigation of the Drag Coefficients of Two Single-engine Supersonic Interceptor Configurations from Mach Number 0.8 to 1.90 to Determine the Effect of Inlet and Engine Locations Variation of the Drag Coefficient with Wind and Wave State Multiphase Flows with Droplets and Particles Drag Coefficients for Spheres in Free

Molecular Flow in O at Satellite Velocities The Variation of the Drag Coefficient in the Marine Surface Layer Due to Temporal and Spatial Variations of the Surface Wind and Sea State Summary of Drag Coefficients of Various Shaped Cylinders Determination of the Hypersonic-continuum/rarefield-flow Drag Coefficient of the Viking Lander Capsule 1 Aeroshell from Flight Data Shock Tube Determination of the Drag Coefficient of Small Spherical Particles Drag coefficients of inert, burning, or evaporating particles accelerating in gas streams Determination of the Hypersonic-continuum/rarefied-flow Drag Coefficient of the Viking Lander Capsule 1 Aeroshell from Flight Data Free Wake Analysis of Hover Performance Using a New Influence Coefficient Method A Two-dimensional Study of the Influence on Target Loading of Numerical Wave Reflections from Transmissive Computational Boundaries University of Iowa Studies in Engineering Drag Associated with Separated Flow Over Two-dimensional V-shaped Notches Under Transonic and Supersonic Conditions The Aeroplane Aero Digest Experiments on Drag of Revolving Disks, Cylinders and Streamline Rods at High Speeds Free-flight Drag Measurements of Rocket- Boosted Models of Two Reentry Body Configurations at Mach Numbers from 4.3 to 0.6 ESDU Engineering Sciences Data: Drag of excrescences, undercarriage, canopy *Robert E. Whitaker Joseph H. Judd Beverly J. Byars Clayton T. Crowe John W. Boring H. Michael Byrne C. F. Heddleson Robert C. Blanchard Bruce P. Selberg Clayton T. Crowe Todd R. Quackenbush Richard E. Lottero Ronald H. Howell Theodore Theodorsen*

the dissipation method is used to obtain estimates for the friction velocity u_{sub} as well as values for the neutral drag coefficient c_{dn} for data collected from a coastal tower off san diego california $c_{\text{sub dn}}$ is found to be independent of the ten meter height windspeed $u_{\text{sub 10}}$ for velocities between 4 9 m sec its value is estimated to be 0 94 or 0 4 1000 which compares well with values by smith 1980 and large and pond 1981 definite trends in $c_{\text{sub dn}}$ with fetch and sea state are also observed drag coefficient estimates are found to be higher for short fetch than for long fetch conditions $c_{\text{sub dn}}$ is also seen to increase sharply just before frontal passages and during sea breeze conditions when the waves are actively growing with the windspeed and wave field reaching equilibrium $c_{\text{sub dn}}$ is found to decrease with time to a smaller and more constant value author

multiphase flow technology especially in the area of gas droplet and gas particle flows is increasingly important in the energy and manufacturing industries pollution control pneumatic transport food processing combustion and development of new materials as well as many other engineering applications will benefit from the fundamental engineering design applications and research in this field written for graduate students and professionals multiphase flows with droplets and particles provides a clear pedagogical approach to the fundamentals of gas particle and gas droplet flows

the drag coefficients for the echo 1 and explorer 24 spherical surfaces in an o environment were experimentally determined over an energy range of 4 to 200 ev the experiment was performed by generating a beam of atomic oxygen ions of the proper energy neutralizing a portion of the beam and then allowing only the neutral o particles to strike a very sensitive torsion balance the momentum transferred to the surface was determined from the deflection of the torsion balance at the lower energies the more intense ion beam had to be used instead of the neutral beam the drag coefficients are found to be slightly greater than 2 at energies corresponding to satellite velocities

data are presented which can be used to predict the aerodynamic drag force caused by cylinders having different cross sectional shapes drag characteristics of circular square rectangular triangular diamond and elliptical cylinders as well as a large range of thickness to chord ratio symmetrical airfoil shapes are presented as a function of reynolds number in addition certain mach number characteristics are discussed

an experimental study was conducted to determine the drag coefficient of inert spherical particles accelerating in a laminar non reacting incompressible continuum flow the reynolds number range which was covered in the study was from 150 to 1700 and particle sizes ranged from 150 microbar to 450 microbar the convective flow behind the shock wave in a shock tube was used to accelerate the particles the particle s diameter and the displacement versus time measurements were obtained using a rotating drum camera in conjunction with an oscillating light source the photographic data the particle density the shock speed and the initial pressure and temperature in conjunction with the normal shock relations were combined to calculate the drag coefficient the drag coefficient is usually considered to be a function only of reynolds number and acceleration modulus however $c_{sub} d$ varies considerably because of particle roughness experiments with hp 295 ball powder whose surface is relatively rough produced results which were as much as 85 per cent higher than the steady state curve with the increase dependent upon the relative mach number of the flow about the particle similar drag coefficient experiments with smooth sapphire balls did not produce the scatter the higher values nor the dependence on relative mach number

this report quantifies the changes in the loading on a target caused by the arrival of artificial numerically induced reflections of waves from the transmissive boundaries of a computational grid several computations were performed using the two dimensional cartesian coordinates mode of the ballistic research laboratory s version of the airblast hull hydrodynamics computer code hull

uses a two step explicit differencing method to solve the inviscid unsteady euler equations a target is simulated in the computational grid by generating aggregates of rigid immobile and impermeable flow field cells the simple transmissive boundaries in hull simulate a zero gradient condition across the boundary for both the pressure and the normal component of velocity simple transmissive boundaries such as these will partially reflect waves that strike them including shock compression and expansion waves the strength of these reflected waves is directly related to the strength of the incident waves these reflected waves then travel back into the computational grid modifying the flow field conditions in the regions through which they pass thereby ending the simulation of free field conditions

summary an experimental investigation concerned primarily with the extension of test data on the drag of revolving disks cylinders and streamline rods to high mach numbers and reynolds numbers is presented a mach number of 2 7 was reached for revolving rods with freon 113 as the medium the tests on disks extended to a reynolds number of 7 000 000 parts of the study are devoted to a reexamination of the von kármán prandtl logarithmic resistance law and the ackeret taylor supersonic drag formula and conditions for their validity the tests confirm in general earlier theories and add certain new results a finding of first importance is that the skin friction does not depend on the mach number of interest also are experimental results on revolving rods at very high mach numbers which show drag curves of the type familiar from ballistics a new result which may have general applicability is that the effect of surface roughness involves two distinct parameters particle size and particle unit density the particle size uniquely determines the reynolds number at which the effect of the roughness first appears whereas the particle unit density determines the behavior of the drag coefficient at higher reynolds numbers beyond the critical reynolds number at which the roughness effect appears the drag coefficient is found to be a function of unit density in the limiting case of particle saturation or a maximum density of particles the drag coefficient remains constant as the reynolds number is increased

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